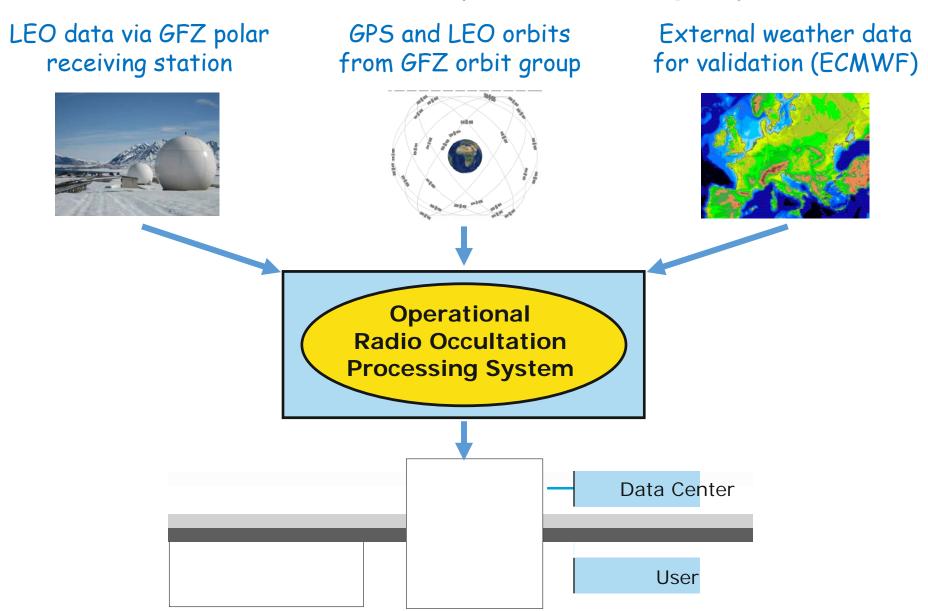
An overview of radio occultation activities at GFZ Potsdam: data processing and applications

Torsten Schmidt, Patrick Schreiner, Florian Zus, Jens Wickert

(GFZ Potsdam)

With contributions from JPL (Byron Iijima et al.)

Radio occultation processing system



GFZ Radio Occultation Missions

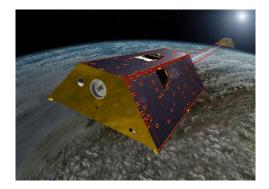




GRACE A/B 2006-2017 setting



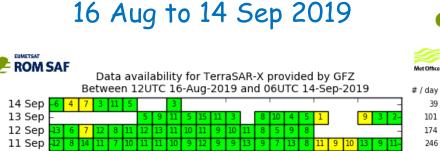
TerraSAR-X since 2008, setting Tandem-X since 2011, rising

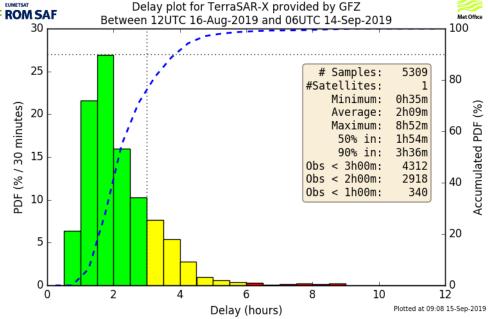


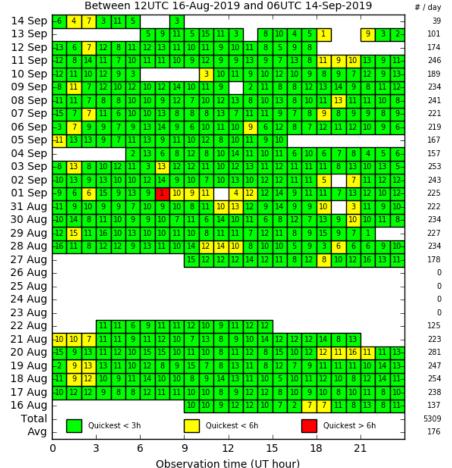
GRACE Follow-on GRACE C/D since 2019

GRACE C, rising

Example: TerraSAR-X data delivery







Mean time delay between onboard measurements and data availability is about 2 hours.

About 90% of all data are delivered within 3 hours.

GRACE Follow-on

Some milestones

- Launched on May 22, 2018.
- Twin satellites follow each other in orbit (in about 490 km) separated by about 200 km.
- Send constantly microwave and laser signals to each other to measure the distance between them. \rightarrow Gravity field estimation \rightarrow Main goal of the mission!
- Summer 2018 problems with IPU on GF2 (switched to second IPU in autumn 2018)
- After some (setting) test ROs from GF2 in summer 2018 no ROs for about one year
- Continuous ROs from GF1 (rising occs) since June 14, 2019
- Updated RO reader provided by JPL on July 25



GRACE-FO Instrument frey Tien Processing Unit

Courtesy of Thomas Meehan and Jeffrey Tien

- Based on TriG-Lite design
- Ka/K band tracking for micron-level intersatellite ranging measurement
- Track up to 12 dual frequency GPS Satellite
 - Real-time PVT (position/velocity/timing)
 - 1-sec pseudorange and phase measurements for post-processed precise orbit determination (POD)
- GPS radio occultation
 - L1CA/L2P, L1CA/L2C, or L1CA/L5
 - Rising radio occultation on the leading satellite
 - Setting radio occultation on the trailing satellite
 - Up to 300 profiles per day per satellite
- Programmable RF downconverter able to receive all L band signal
 - Reconfigurable Software allows additional capabilities (e.g. Galileo GNSS RO) to be added later

GPS radio occultation antenna

Zenith pointing GPS antenna







Instrument Processing Unit (IPU)

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RO reader (from JPL) output

Courtesy of Angela Dorsey, JPL

Currently tracking L1CA, L2P, L5 (prn30, prn32), L2C RO is currently disabled.

- I_Sig1 In-phase correlation product for signal 1 (L1 CA) over 1/numPoints seconds
- Q_Sig1 Quadrature correlation product for signal 1 (L1 CA) over 1/numPoints seconds
- I_Sig2 In-phase correlation product for signal 2 (L2P, L2C, or L5) over 1/numPoints seconds
- Q_Sig2 Quadrature correlation product for signal 2 (L2P, L2C, or L5) over 1/numPoints seconds
- phaseModelSig1 CA model (cy)
- phaseModelSig2 P2/L5 model (cy)
- pseudoRangeSig1 CA pseudorange (m)
- pseudoRangeSig2 P2/L2C/L5 pseudorange (m)

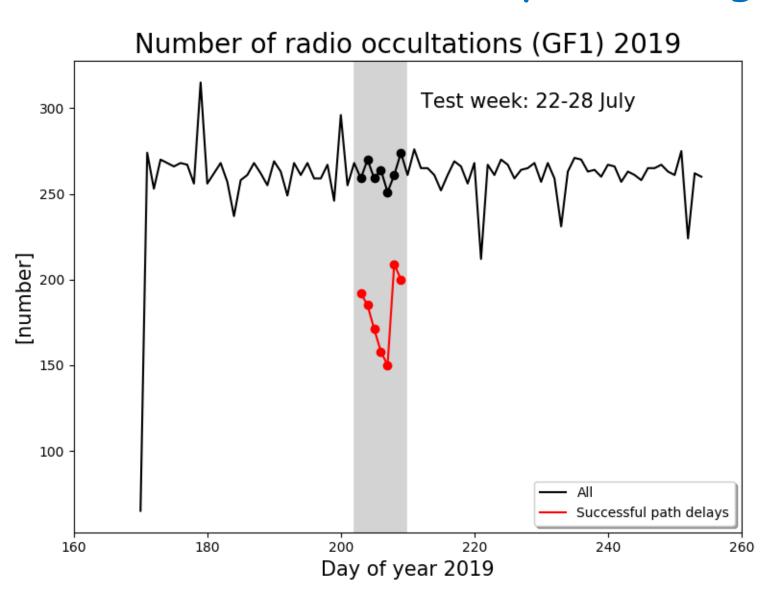
In CL I_Sig1 = 50Hz amplitude for CA, Q_Sig1=0, I_Sig2 = 50Hz amplitude for P2/L2C/L5, Q_Sig2=0

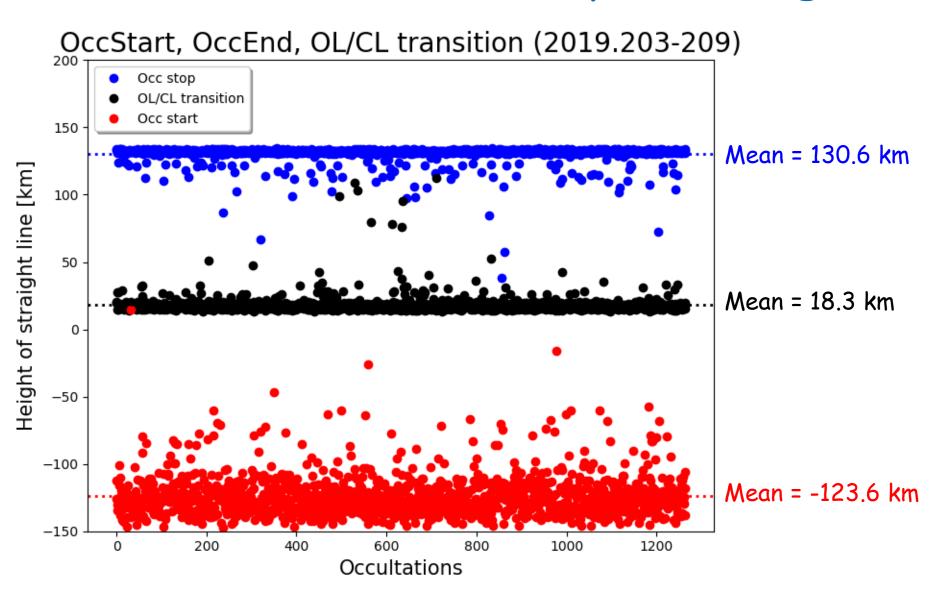
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617199765 0.427297960 OL prn9 -7612 11941 -4099 -2851 61507796.511785 89457943.403831 29171682.617979 29171700.762967 617199765 0.447297564 OL prn9 -10817 9227 -5143 -6441 61507172.199523 89457456.939741 29171682.617979 29171700.762967 617199765 0.467297168 OL prn9 -13249 6762 -5934 -14352 61506547.885446 89456970.586664 29171682.617979 29171700.762967 617199765 0.487296772 OL prn9 -14703 2141 -6845 -21014 61505923.569677 89456484.070308 29171682.617979 29171700.762967 617199766 -0.500 CL prn9 14911 0 451 0 61505527.019257 89456174.785529 29165741.606538 29165750.345726 617199766 -0.480 CL prn9 15111 0 93 0 61504902.646376 89455688.454518 29165741.606538 29165750.345726 617199766 -0.460 CL prn9 15571 0 2225 0 61504278.256473 89455202.088007 29165741.606538 29165750.345726 617199766 -0.440 CL prn9 15854 0 1211 0 61503653.873963 89454715.542321 29165741.606538 29165750.345726
```

OL:
$$TotalPhase = PhaseModelSig + \frac{atan2(Q_{Sig} * B, I_{Sig} * B)}{2\pi} + C_n$$

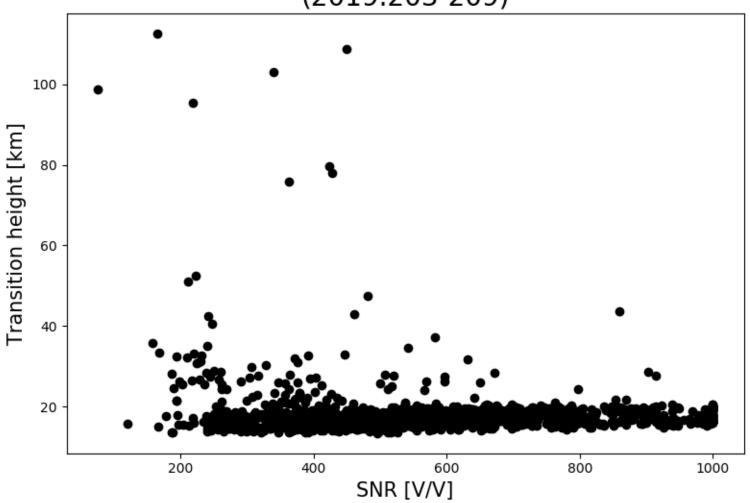
CL: TotalPhase = PhaseModelSig

B: navbit correction C_n : phase unwrapping





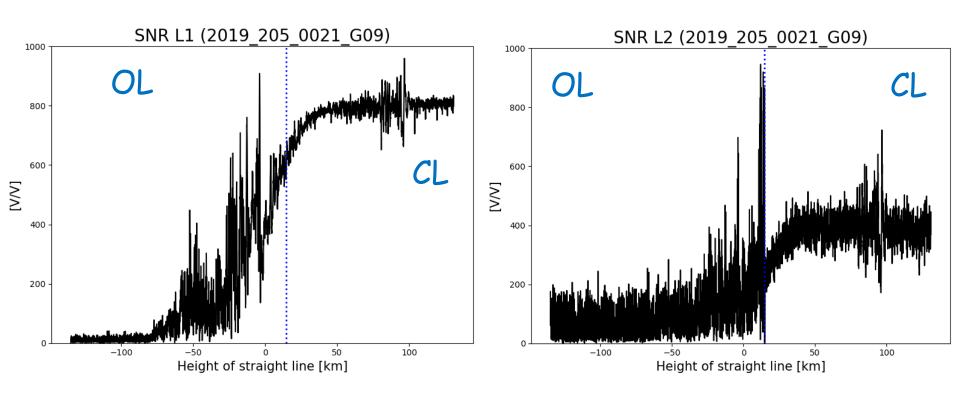
SNR L1 vs. OL/CL transition height (2019.203-209)

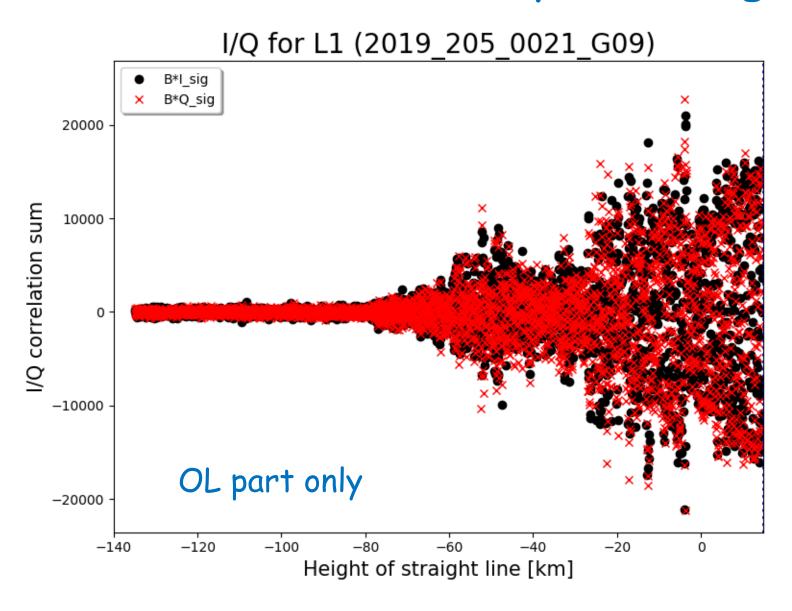


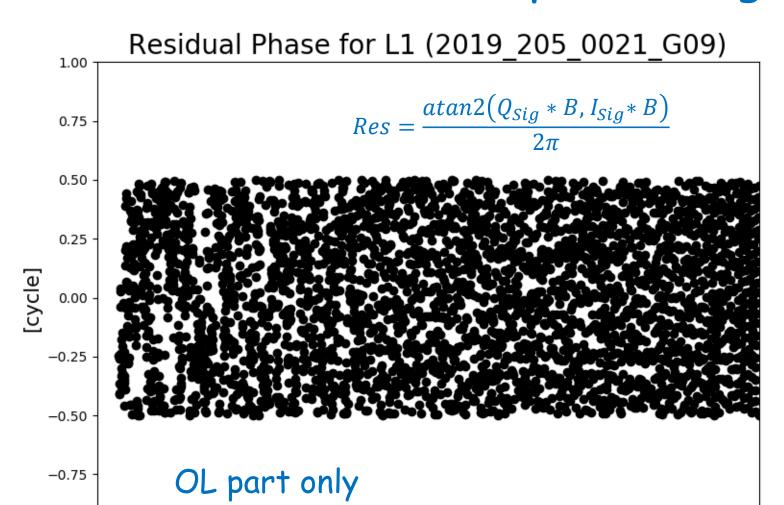
SNR

OL:
$$SNR = Factor * \sqrt{(I^2 + Q^2)}$$

CL: SNR = Factor * Ampl







-80

-60

Height of straight line [km]

-40

-20

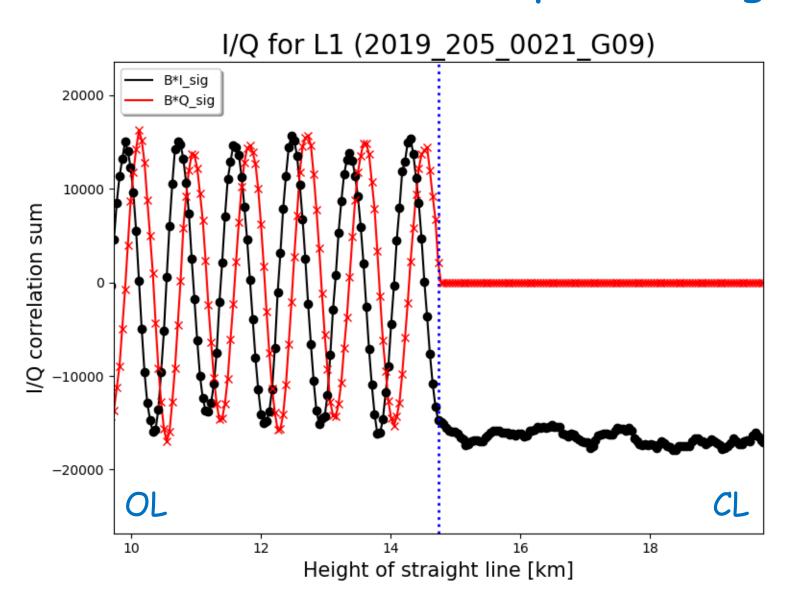
0

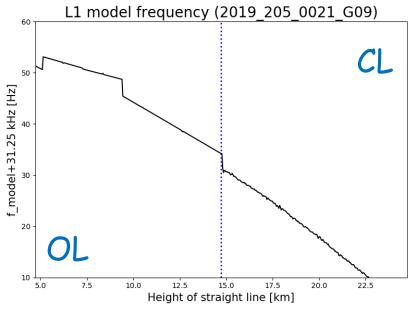
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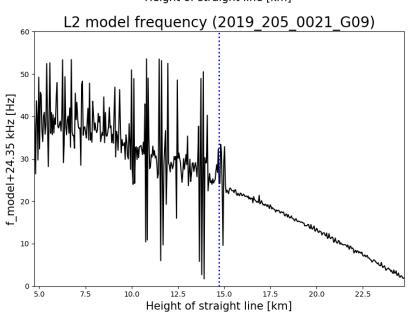
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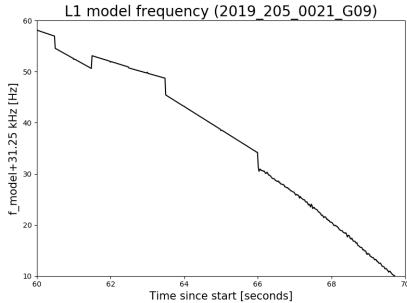
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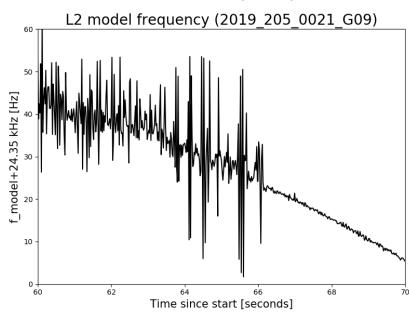
-100

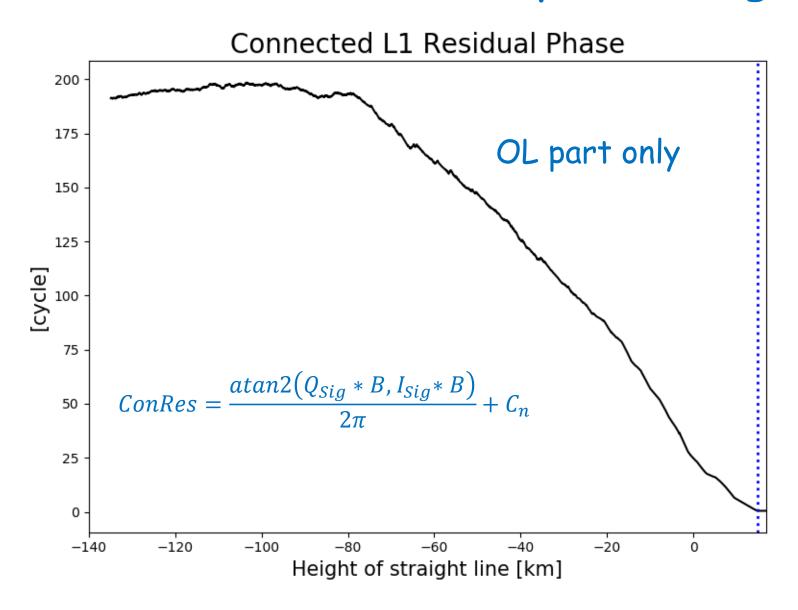




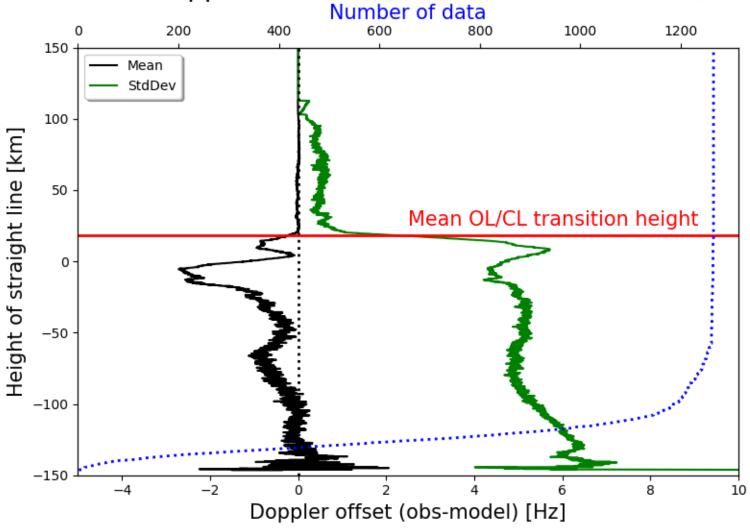


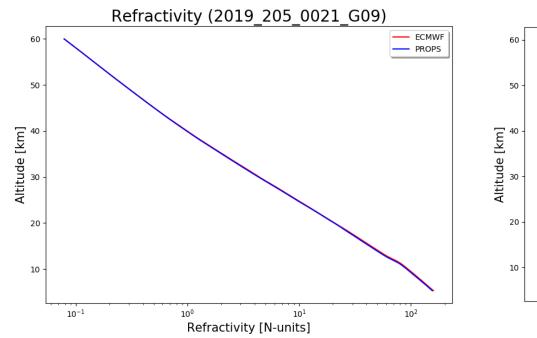


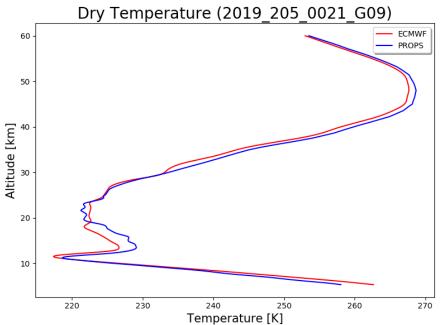




L1 Doppler offset obs-model (2019.203-209)







to be continued

GRACE-FO: Next steps

- Command to lower the OL/CL transition to -15 km (as for TSX/TDX)
- Software update
- The new IPU software is expected to fix L2C RO (L2C RO is currently disabled.)
- The SW update will be in November for GF1 and perhaps December for GF2.
- Probably RO on GF2 IPU will be turned on.

Other activities at GFZ: New missions

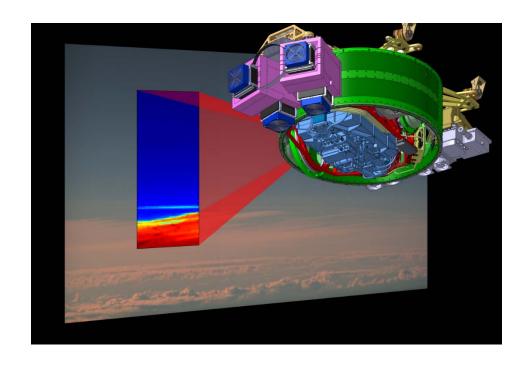
AtmoSat

Highest ranked proposal on the "national roadmap" for big infrastructure projects

Together with Research Center Jülich, KIT Karlsruhe, DLR and other

With RO (TriG)

In the waiting loop



Focus is middle atmosphere (5-100 km) Observables:

Temperature, H2O, HDO,O3, CH4, N2O, CFC-11, CFC-12, HCFC-12, SF6, HNO3, N2O5, ClONO2, HO2NO2, PAN, C2H6, H2CO, NH3, aerosol and cirrus properties

Resolution: $1 \text{ km} \times 25 \text{ km} \times 50 \text{ km}$ (global)

Other activities at GFZ: New missions

PRETTY

Cubesat for precise altimetry using navigation satellites



- Small satellite, cubesat (10*10*30cm³)
- ESA mission, consortium led by RUAG with IEEC and UoO
- GNSS-Reflectometry and Space Weather
- Precise altimetry using grazing geometries
- Unique: Direct correlation of direct and reflected GNSS signals
- Launch expected for 2021

Summary

- GFZ continues with operational RO processing for TSX/TDX
- GRACE-FO RO data available since this summer
 - High OL-CL transition height (compared to TSX/TDX)
 Will be changed soon, expecting big impact
 - Software updates in Nov/Dec with expected improvements (L2C RO availability)
 - Continue with data analysis
- New RO mission(s)

Thank you for your attention!