

GRAS SAF Open Loop Workshop
Helsingør, Denmark
June 6-8, 2005

DMI Technical Report 05-11

ISSN: 1399-1388

Kent B. Lauritsen and Frans Rubek, editors



GRAS Open Loop Tracking

Jacob Christensen
Saab Ericsson Space
jacob.christensen@space.se

GRAS Open Loop Tracking

Content:

- Justification for Open Loop Tracking
- Open Loop Tracking & Doppler Model
- Open Loop Data Format & Usage
- Questions / Answers

Jacob Christensen (jacob.christensen@space.se)

GOLW
June 6 – 8, 2005

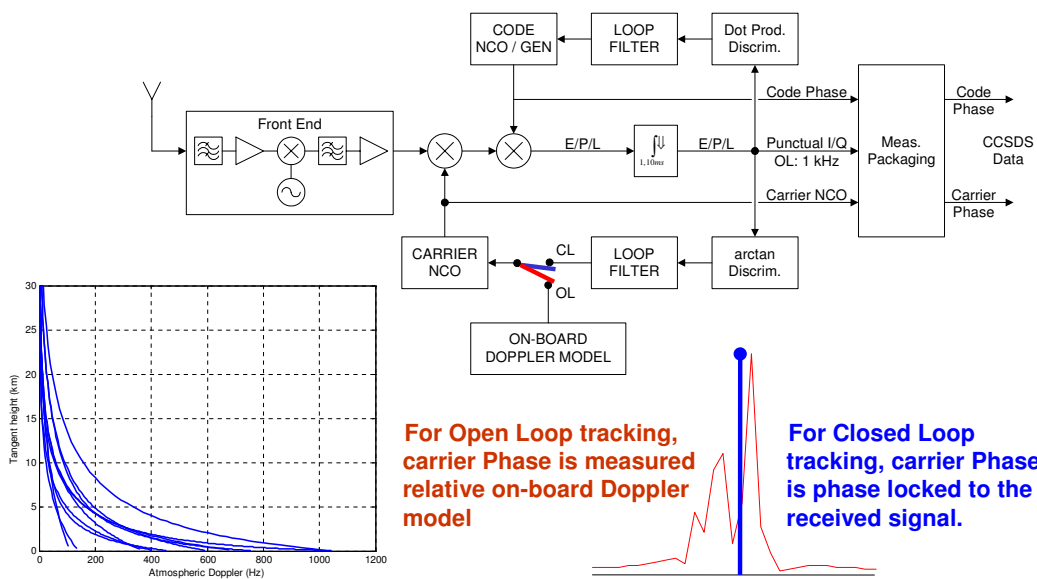
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

1

Open Loop Tracking



GOLW
June 6 – 8, 2005

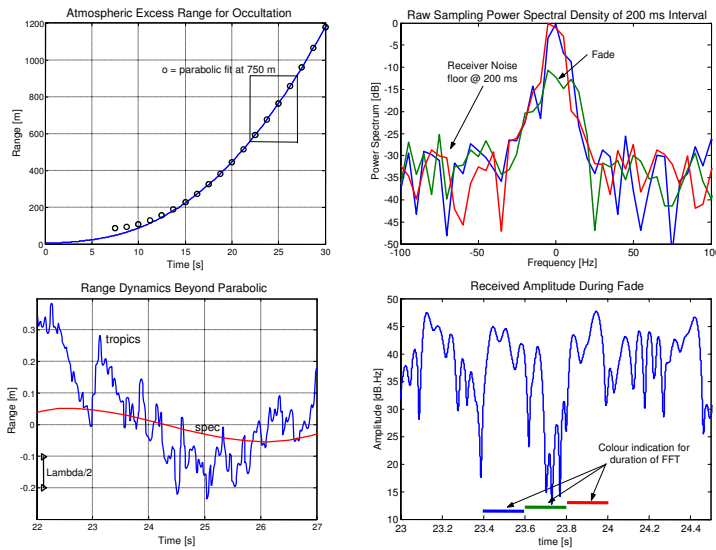
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

2

Atmospheric Multipath Path Signal Sample



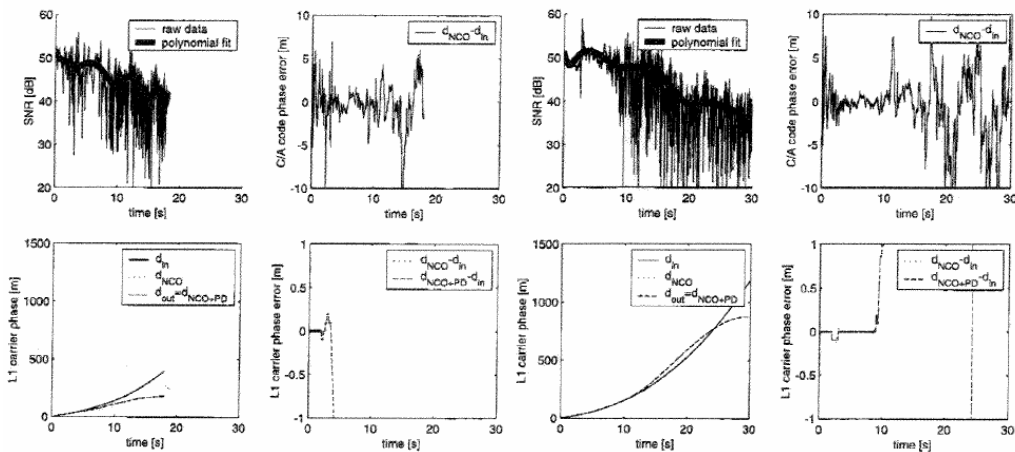
GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Code and Carrier Closed Loop Tracking Simulation



- **Code Tracking is maintained**
- **Carrier Tracking is lost**
- **Carrier Measurements Supported by Open Loop Tracking**

Data: High resolution Radiosonde data forward modeled by UCAR

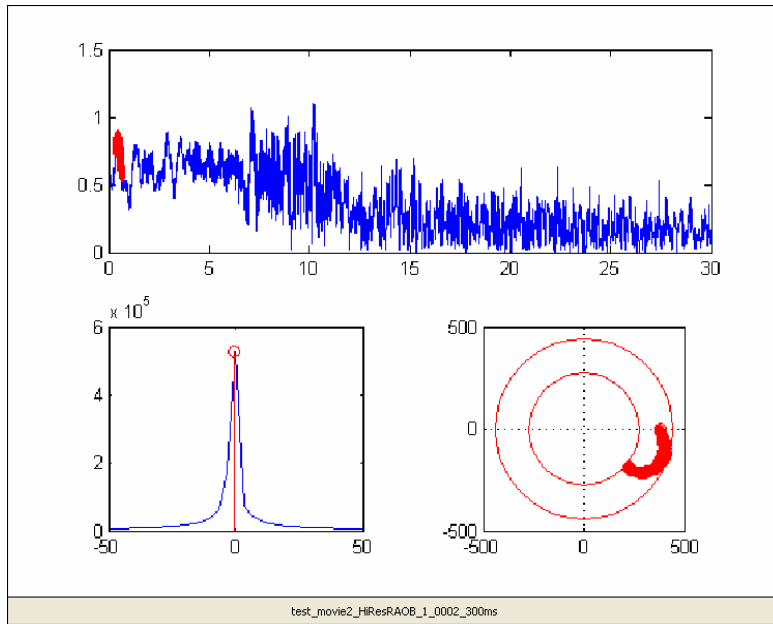
GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

300 ms Spectrum & Residual I/Q simulation



GOLW
June 6 – 8, 2005

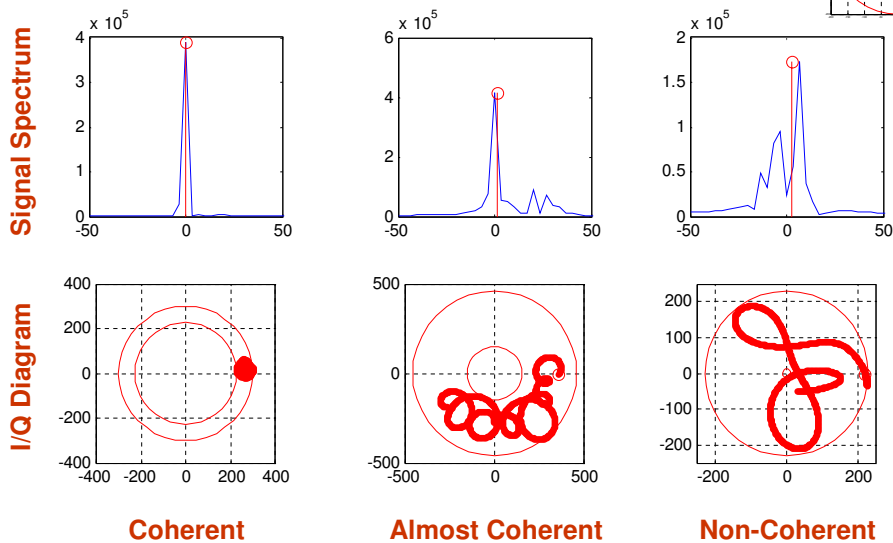
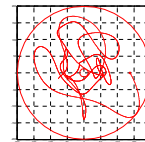
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Coherency

(From hi resolution radiosonde data)



GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.

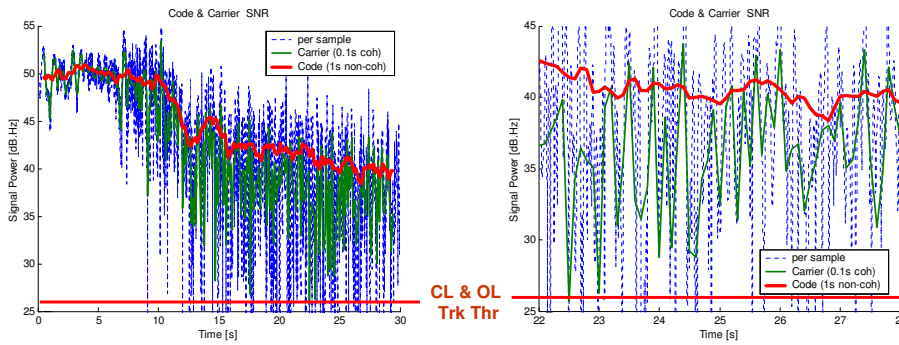


Saab Ericsson Space

Code & Carrier SNR

For severe atmospheric conditions:

- Carrier tracking will be lost due to lack of coherency (low SNR & dynamics)
- Code tracking is non-coherent and more resistant to “atmospheric dynamics”
- Open loop tracking rely only on code tracking and measures the carrier relative an onboard Doppler model.



GOLW
June 6 – 8, 2005

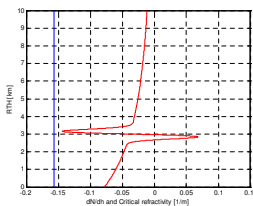
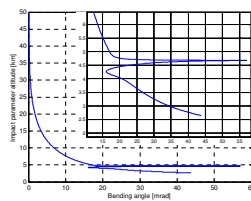
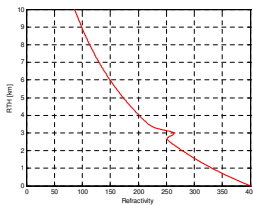
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.

Data: HI-resolution Radiosonde data (UÇAR)

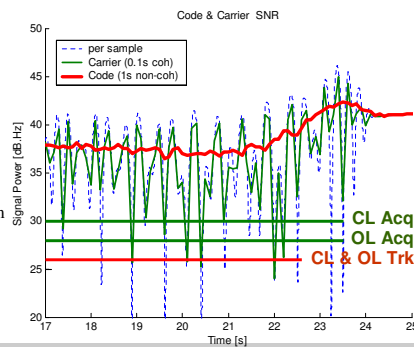
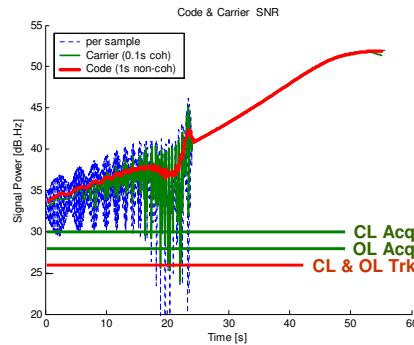


Saab Ericsson Space

Refractivity Bump Simulation In “Dense Wet” Atmosphere



Refractivity at surface: $N_0 = 300$
 Scale height: $H_0 = 7.9$ km
 Wet refract. at surface: $N_w = 100$
 Wet scale height: $H_w = 2.5$ km
 Bump strength: $N_b = 28$
 Bump height: $h_b = 3$ km
 Bump width: $w_b = 0.2236$ km



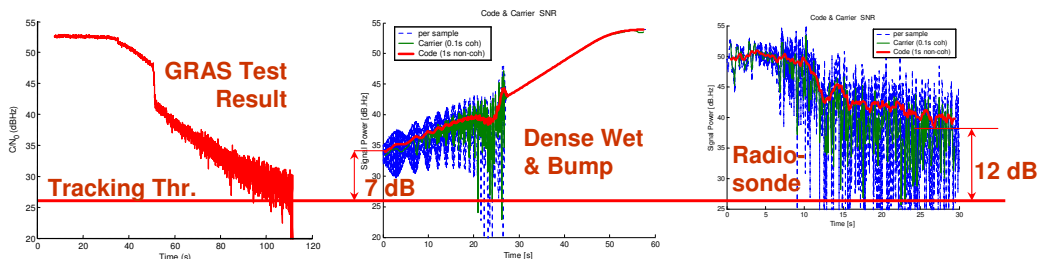
GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Signal tracking conditions



Real life versus GRAS Spec/Test:

- Closed Loop Tracking will be more intermittent.
- Open Loop Tracking will have more margin and be very robust.
- Open Loop Data will be more significant for real signals
- Retune instrument parameters to optimize open loop tracking for rising occultations. (SLTA_L2 & SLTA_A)

GOLW
June 6 – 8, 2005

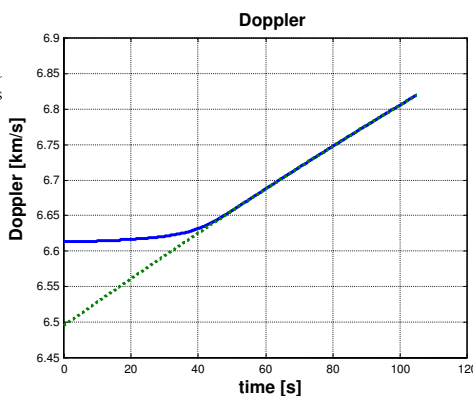
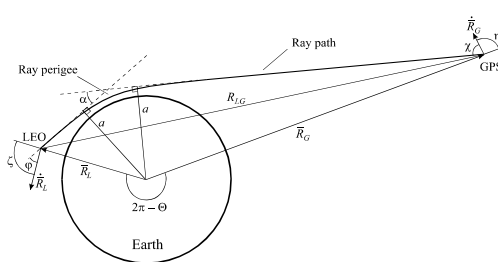
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

9

Doppler Model (Patented)



- Measured Doppler approximated as:
 - Line of sight at high altitude
 - Constant impact parameter altitude above WGS 84 at low altitudes.
- Nominal Ray Impact Points (NRIP) modeled on-board based on Straight Line Tangent Altitude (SLTA)

GOLW
June 6 – 8, 2005

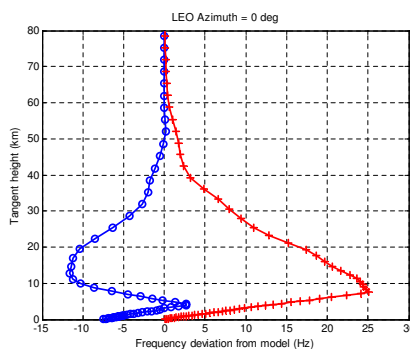
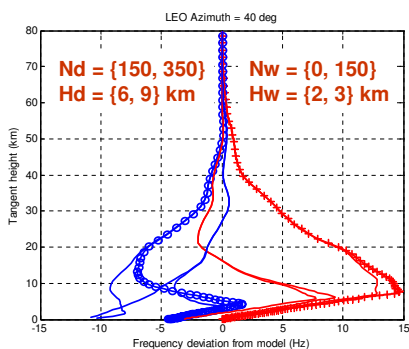
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



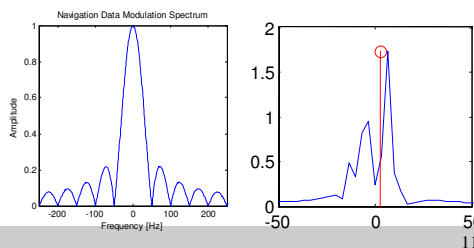
Saab Ericsson Space

10

Doppler Model Accuracy



- Doppler Model residual: $< \pm 25$ Hz
- Atmospheric modulation: up to $\sim \pm 25$ Hz
- I/Q bandwidth: 50 - 100 Hz without nav data
- Nav. Data modulation: ± 150 Hz @ -20 dB



GOLW
June 6 – 8, 2005

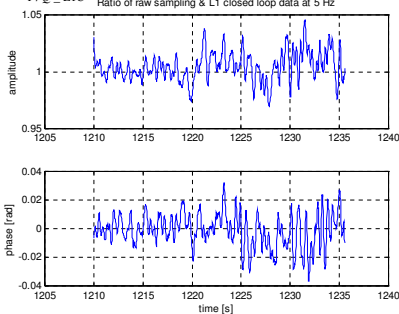
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



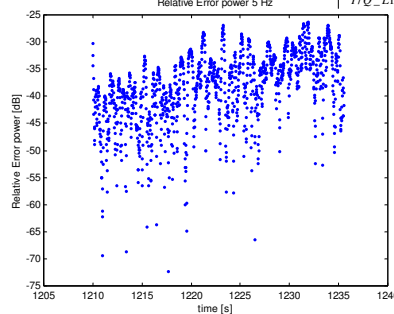
Saab Ericsson Space

Open to Closed Loop Alignment

$$R_{I/Q} = \frac{V_{I/Q_RS}}{V_{I/Q_LIC}}$$



$$\epsilon_R = \frac{|V_{I/Q_RS} - V_{I/Q_LIC}|}{|V_{I/Q_LIC}|}$$



Measured OL & CL Data:

$$V_{Loop}^{RS} = (I_{RS} + j \cdot Q_{RS}) \cdot e^{j\phi_{RS_NCO}}$$

$$V_{Loop}^{LIC} = (I_{LIC} + j \cdot Q_{LIC}) \cdot e^{j\phi_{LIC_NCO}}$$

Evaluation Products:

$$V_{I/Q_RS} = (I_{RS} + j \cdot Q_{RS}) \cdot e^{j(\phi_{RS_NCO} - \phi_{LIC_NCO})}$$

$$V_{I/Q_LIC} = (I_{LIC} + j \cdot Q_{LIC})$$

The relative error between open & closed loop measurement data is better than -20 dB.

GOLW
June 6 – 8, 2005

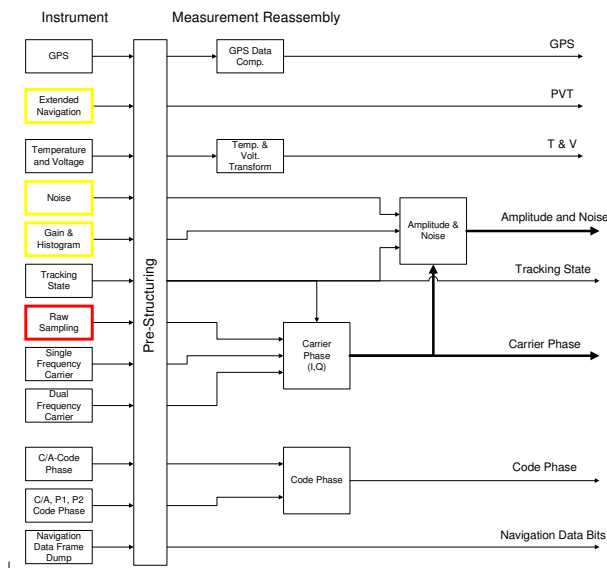
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

GRAS Measurement Data

- "Raw Sampling" packet provide:
 - Doppler Model Phase
 - Raw I/Q samples at 1 kHz
 - Time tagged with IMT, (Instrument Measurement Time)
- Relation between IMT and UTC time from:
 - "Extended Navigation" packet
- Amplitude Scaling of raw I/Q data from:
 - "Gain and Histogram" packet
 - "Noise" packet



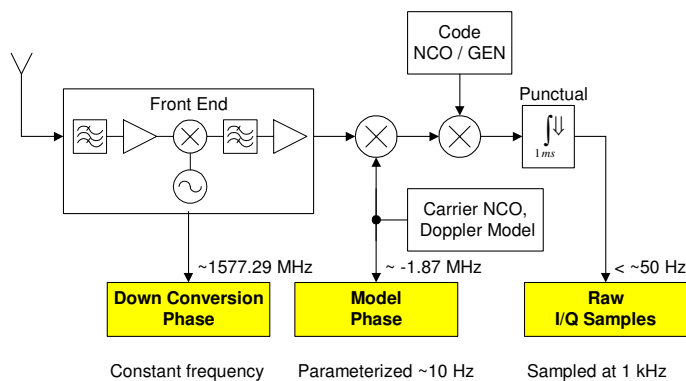
GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Open Loop Data Components



$$V_{Ant}(t) = \frac{1}{\sqrt{G_k}} \cdot e^{j\phi_{DC}^{L1}(t)} \cdot e^{j\phi_{HP}^{L1}(t)} \cdot (I(t) + jQ(t))$$

Syst. Const.
Open Loop Measurement Data

$$\phi_{DC}^{L1} = \frac{7571}{24} \cdot f_{USO} \cdot (t - t_0) + \phi_{DC}^{L1}(t_0); \quad f_{USO} = 5.0MHz \pm 0.1Hz$$

The measured signal (V_{ant}), represent the GPS signal as received at the antenna and includes the GPS carrier at 1575.42 MHz (nom).

GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.

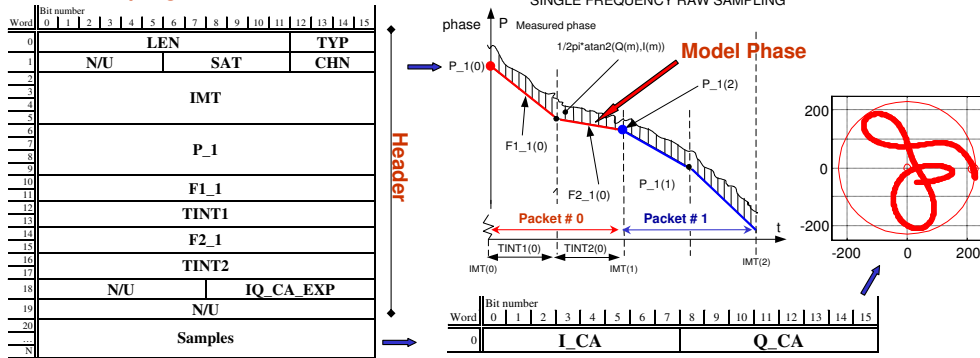


Saab Ericsson Space

Open Loop Measurement Data

(Packet Type: 9 “Occultation Raw Sampling”)

“Raw Sampling” Data Packet:



- Binary Data format described in: P-GRM-ICD-0008-SE
“Measurement Data Interface Control Document”
- Data Interpretation described in: P-GRM-SPC-0036-SE
“Measurement Data Interpretation and Description”

GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Open Loop Measurement Data (2)

SUB HEADER:

Label	Symbol	Size [bits]	Resolution [units]	Range [units]	Unit	Content
IMT	$T_{IMT}(n)$	64	1	$0-2^{64}$	Core_clk	IMT of first phase measurement in packet number n.
P_1	$\phi_P^{L1}(n)$	64	2^{-12}	$\pm 2^{47}$	Cycles	L1 NCO phase at $T_{IMT}(n)$ in packet n in units of NCO phase resolution.
F_1	$F_{F1}^{L1}(n)$	32	1	$\pm 2^{31}$	f _{NCO_Res}	Reference frequency of L1 carrier for segment 1 and 2 of packet number n in units of NCO frequency resolution.
F_2	$F_{F2}^{L1}(n)$	32	1	$\pm 2^{31}$	f _{NCO_Res}	
IQ_CA_EXP	$E_{CA}(n)$	8	1	0-255	LSB	Exponent of L1 carrier I/Q samples in packet n.
TINT1	$T_{int}(n)$	32	1	$0-2^{32}$	Core_clk	Total integration time for all measurements in segment 1 and 2 of packet number n.
TINT2	$T_{int}(n)$	32	1	$0-2^{32}$	Core_clk	Total integration time for all measurements in segment 1 and 2 of packet number n.
CHN	Ch	4	1	8-11	LSB	Measurement channel id. GVA: 8,9; GAVA: 10,11
LEN	Len(n)	12	1	$0-2^{11}$	byte	Length of packet number n in bytes

BODY DATA:

Label	Symbol	Size [bits]	Resolution [units]	Range [units]	Unit	Content
I_CA	$I_{CA}(m)$	8	$2^{IQ_exponent}$	$\pm 2^{IQ_exponent+7}$	AGGA Integration Counts	m'th I and Q data of L1 carrier amplitude in terms of normalised counts of the C/A punctual integrate and dump.
Q_CA	$Q_{CA}(m)$	8	$2^{IQ_exponent}$	$\pm 2^{IQ_exponent+7}$	AGGA Integration Counts	

GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Open Loop Data Interpretation

Complex signal at antenna:

$$V_{Ant}(t(n, m)) = \sqrt{G_k^{-1}(ch)} \cdot (I_{CA}(n, m) + jQ_{CA}(n, m)) \cdot e^{j\phi_{DC}^{L1}(t)} \cdot e^{j\phi_{HP}^{L1}(n, m)}$$

Receiver Gain (G_k):

From "Gain & Histogram" and "Noise" packets according to MDID.

Sample time (IMT):

$$t(n, m) = t_{IMT}(n) + \frac{\tau_{Int1}(n) + \tau_{Int2}(n)}{M_{samp}(n)} \cdot \left(\frac{1}{2} + m\right) [s]$$

where

$$t_{IMT}(n) = T_{IMT}(n) \cdot \tau_{Core}$$

$$\tau_{Int1}(n) = T_{Int1}(n) \cdot \tau_{Core}$$

$$\tau_{Int2}(n) = T_{Int2}(n) \cdot \tau_{Core}$$

Down conversion Phase:

$$\phi_{DC}^{L1}(t) = \frac{7571}{24} \cdot f_{USO} \cdot (t - t_0) + \phi_{DC}^{L1}(t_0)$$

$$n: \text{Packet number}; \quad \tau_{Core} = \frac{1}{f_{Core}}; \quad f_{Core} = \frac{113}{20} f_{USO}; \quad f_{USO} = 5MHz \pm 0.1Hz; \quad M_{samp}(n) = \frac{Len(n) - 40}{2}$$

Header Model Phase [cycles]:

First segment :

$$\phi_{HP}^{L1}(n, m) = \phi_p^{L1}(n) + f_{F1}^{L1}(n) \cdot \frac{\tau_{Int1}(n) + \tau_{Int2}(n)}{M_{samp}(n)} \cdot \left(\frac{1}{2} + m\right)$$

for $m = \{0, 1, 2, \dots, [M_1 - 1]\}$

Second segment :

$$\phi_{HP}^{L1}(n, m) = \phi_p^{L1}(n) + f_{F1}^{L1}(n) \cdot \tau_{Int1}(n) + f_{F2}^{L1}(n) \cdot \frac{\tau_{Int1}(n) + \tau_{Int2}(n)}{M_{samp}(n)} \cdot \left(\frac{1}{2} + m - M_1\right)$$

$$\text{for } m = \{M_1, \dots, (M_{samp} - 1)\}$$

$$\text{Where: } M_1 = \text{Round}\left(\frac{\tau_{Int1} \cdot M_{samp}}{\tau_{Int1} + \tau_{Int2}}\right) \text{ and } f_{Fx}^{L1}(n) = F_{Fx}^{L1}(n) \cdot \frac{f_{Core}}{12 \cdot 2^{28}}, \quad x = \{1, 2\}$$

GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

17

Open Loop Data Interpretation

Complex signal at antenna:

$$V_{Ant}(t(n, m)) = \sqrt{G_k^{-1}(ch)} \cdot (I_{CA}(n, m) + jQ_{CA}(n, m)) \cdot e^{j\phi_{DC}^{L1}(t)} \cdot e^{j\phi_{HP}^{L1}(n, m)}$$

Receiver Gain (G_k):

Calculation described in MDID

Sample time (IMT):

$t(n, m)$ calculated from:

- IMT
- TINT1
- TINT2
- LEN

Down conversion Phase:

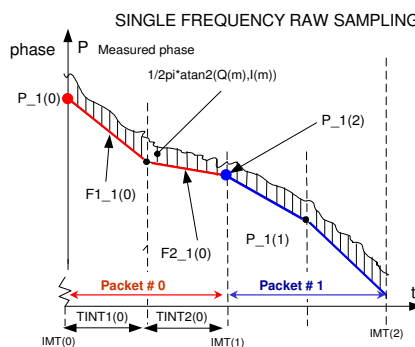
$$\phi_{DC}^{L1}(t) = \frac{7571}{24} \cdot f_{USO} \cdot (t - t_0) + \phi_{DC}^{L1}(t_0)$$

$$n: \text{Packet number}; \quad m: \text{Sample number};$$

Header Model Phase [cycles]:

$\phi_{HP}^{L1}(n, m)$ calculated from:

- P_1
- F1_1
- F2_1
- TINT1
- TINT2
- LEN



Keep Phase & I/Q as separate components

GOLW
June 6 – 8, 2005

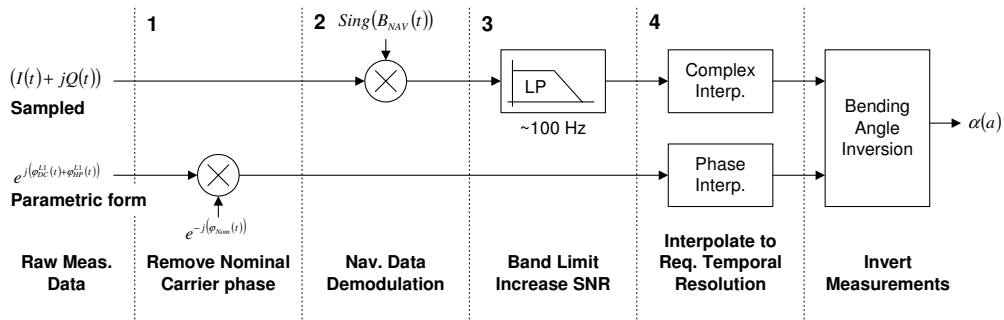
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

18

Using Open Loop Data



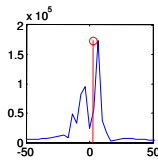
- 1:**
 - Remove nom GPS freq => $f < -50$ kHz
 - Remove mean/LOS Doppler => $f < -2$ kHz
 - Coordinate with clock corr.
 - Coordinate with Inversion algorithm
- 2:**
 - Remove Nav. Data Mod. to eliminate π -ambiguities.
- 3:**
 - Maximize SNR before any non-linear processing is applied. ~ -5dB @ Trk. Thr. & 1 kHz
- 4:**
 - I/Q data is over sampled after band limiting. => Low order interp OK
 - Interpret phase to avoid aliasing. i.e. $\Delta\phi > 2\pi$ per sample

GOLW
June 6 – 8, 2005

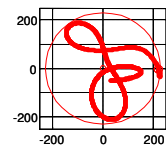
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



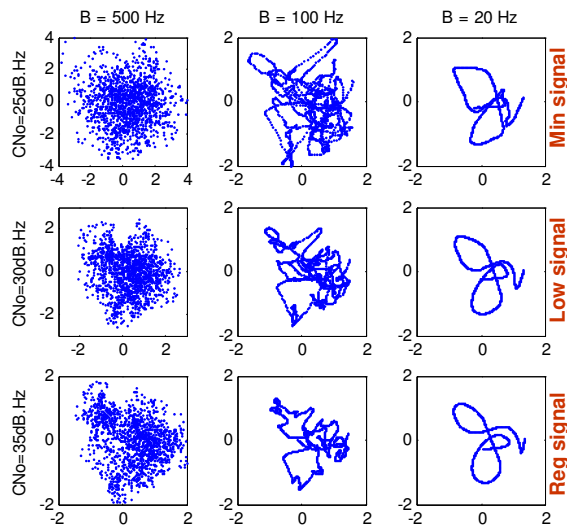
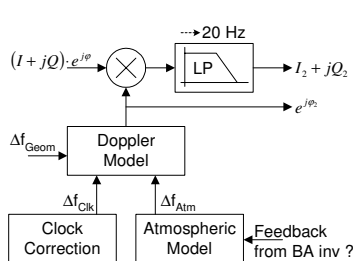
Saab Ericsson Space



Band Limiting



- Coordinate filtering with needs from and implementation of BA. Inversion
- Account for Noise Response of all non-linear operations



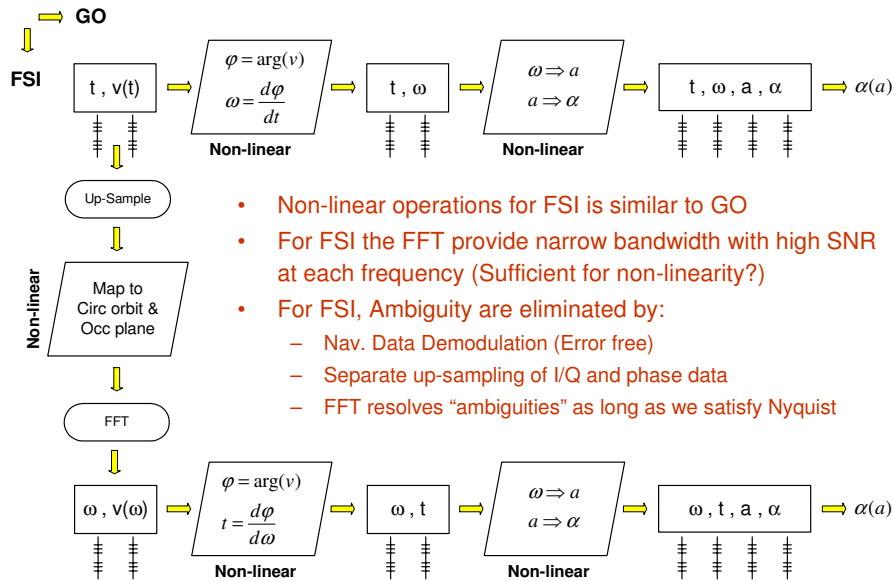
GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Non-linear operations for GO & FSI Inversions



GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.

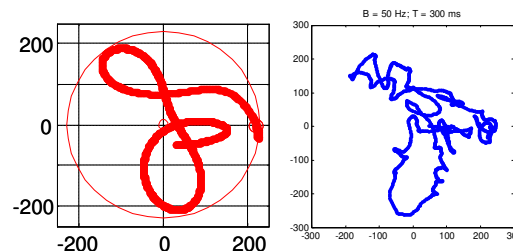


Saab Ericsson Space

21

Atmospheric or Navigation data modulation?

- Presence of Navigation Data Modulation introduce π -ambiguities in the measurement data that needs to be removed.
- The spectrum of the 50 Hz Nav. data modulation is often similar to the spectrum introduced by the “atmospheric modulation”
- Zero crossing from “atmospheric modulation” and navigation modulation looks similar.
- Navigation data modulation is not easily separated from the measured noise polluted signal.



GOLW
June 6 – 8, 2005

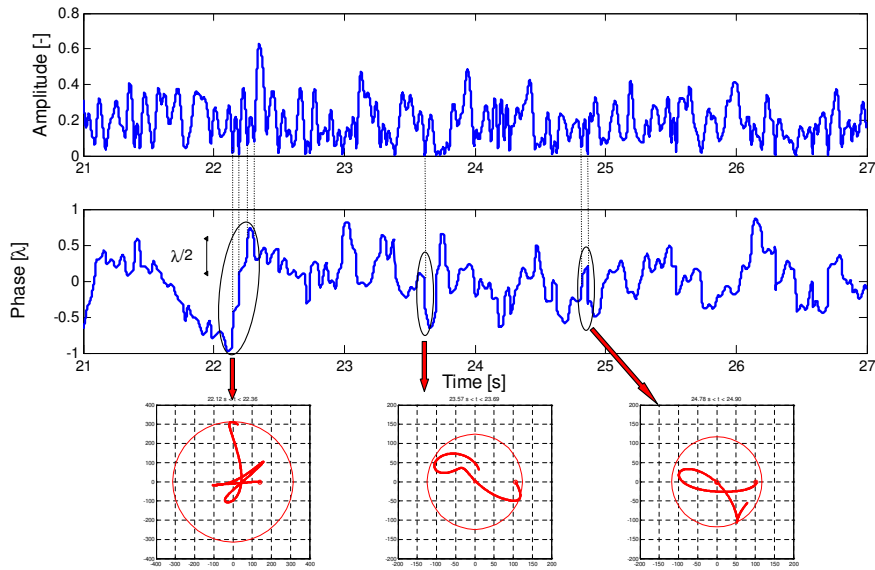
This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

22

Nav Data Demodulation



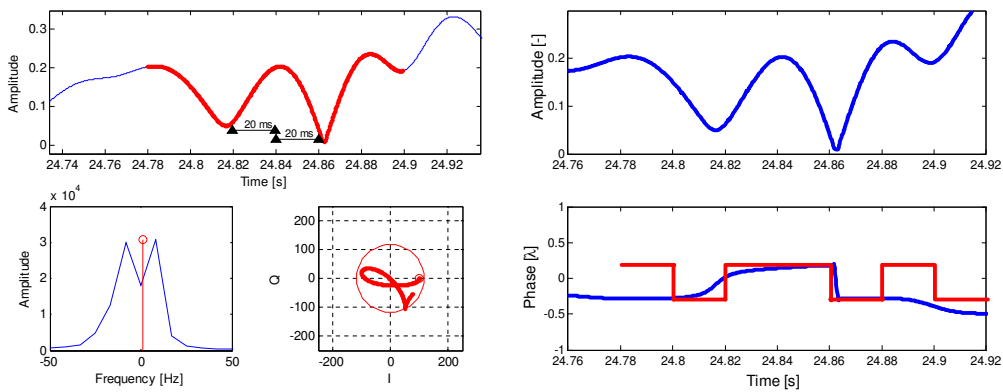
GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Nav Data Demodulation



- Atmospheric modulation or Navigation data modulation?

GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

Navigation Data Demodulation

$$V_{Am}(t) = (I_{RS}(t) + j \cdot Q_{RS}(t)) \cdot \text{sign}(NAV(t)) \cdot e^{j \cdot \phi_{RS}(t)}$$

- Knowledge of Navigation Data Modulation (raw bits) are needed to retrieve atmospheric information from the measured signal.
- Option 1: Bring in navigation data from external source:
 - Retrieve from GPS control segment (availability?)
 - Retrieve from other measurements, e.g. fiducial stations etc.
- Option 2: Estimate navigation data from GRAS measurements:
 - Most of the information content is known from current ephemeris.
 - Use repeatability of navigation modulation and benefit from zenith data retrieved ~15 min before/after the occultation. (Turn on "Navigation Message Dump" by macro command)
 - Minimize dependency on navigation data bit estimations from occultation data.

GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

25

Questions / Answers

- Frequency model used in OL: Climatic, NWP, fly-wheeling? What is the expected accuracy? There are comparisons between the model implemented and actual NWP models?
 - **Doppler model: Doppler towards NRIPA calculated. NRIPA from look-up table against SLTA.**
 - **Accuracy: ~±25 Hz**
- It is possible to use simultaneously two links with one occulting satellite and two different models? The second model could be the standard with a bias in delay or in frequency?
 - **No**
- Could we profit from the additional correlation sums obtained in the early and late arms?
 - **E/L products are not reported to ground**
- Integration times in CL and OL
 - **Clarify integration time.**
 - **For measurement reporting CL: 20 ms (50 Hz); OL: 1 ms (1 kHz)**
- Pre-detection bandwidth
 - **~20 MHz**

GOLW
June 6 – 8, 2005

This document or software is confidential to Saab Ericsson Space AB and must not:
a) be used for any purpose other than those for which it was supplied;
b) be copied or reproduced in whole or in part without the prior written consent of Saab Ericsson Space AB;
c) be disclosed to any third party without the prior written consent of Saab Ericsson Space AB.



Saab Ericsson Space

26

Questions / Answers (2)

- Expected SNR differences between OL and CL
 - **Identical C/No. I.e. SNR power difference = f_{CL}/f_{OL}**
- Removal of the navigation bit: wipe-off or sensitivity assistance
 - **Clarify / Discuss**
- Sampling rate of the observables in CL and in OL
 - **Default & recommended: CL: 50 Hz; OL: 1kHz.**
- Number of correlators used in the estimation of the peak amplitude
 - **Punctual correlation reported.**
- Process chain for the OL: from the RF signal to the level 1a products
 - **Answered or Discuss?**
- Switching from CL to OL: Does the OL model match the CL information?
 - **Yes.**
- Are expected other products in addition to those termed level 1a?
 - **Clarify / Discuss**

